EASA	AIRWORTHINESS DIRECTIVE		
	AD No.: 2012-0197		
	Date: 25 September 2012		
	Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.		
This AD is issued in accordance with EC 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].			
Design Approval I	Design Approval Holder's Name: Type/Model designation(s):		
SOLO KLEINMOTOREN GmbH		2350 C engines	
TCDS Number: Germany <u>Kennblatt Nr. 4603</u>			
Foreign AD: Not applicable			
Supersedure: None			
	r		
ATA 72	Engine – Propeller Shaft – Inspection / Replacement		
Manufacturer(s):	Solo Kleinmotoren GmbH		
Applicability:	Applicability: Model 2350 C engines, all serial numbers, if a Part Number (P/N) 20 31 21 propeller shaft is installed.		
	The affected engines are onl powered sailplanes.	y installed on DG-Flugzeugbau Model DG-1000T	
Reason:	Reason: Two reports have been received of a broken P/N 20 31 211 propeller shaft Solo 2350 C engine. The results of the investigation showed that the failure were due to a material defect, most likely caused by a manufacturing error.		
	and detachment of the prope	d and corrected, could lead to failure of the shaft eller from the aeroplane, which, depending on the in reduced control of the aeroplane, or injury to	
	(magnetic particle or dye per	pove, this AD requires a one-time inspection netrant) of the affected propeller shafts to detect idings, replacement of the propeller pulley rviceable module.	
Effective Date:	09 October 2012		

TE.CAP.00110-003 © European Aviation Safety Agency. All rights reserved. Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously:	
	(1) Within 25 flight hours or 6 months, whichever occurs first after the effective date of this AD, remove the complete propeller pulley assembly from the engine and inspect the transition region of the P/N 20 31 211 shaft in accordance with the instructions of Solo Kleinmotoren GmbH Technische Mitteilung (TM) / Service Bulletin (SB) Nr. 4603-13.	
	(2) If, during the inspection as required by paragraph (1) of this AD, cracks are detected, before release to service of the engine, replace the propeller pulley assembly (module) with a serviceable module.	
	(3) Within 30 days after removal of a propeller pulley assembly (module) from an engine, as required by paragraph (2) of this AD, send the removed module to Solo Kleinmotoren for investigation.	
Ref. Publications:	Solo Kleinmotoren GmbH TM/SB Nr. 4603-13 dated 17 August 2012. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 	
	 Based on the required actions and the compliance time, EASA have decided to issue of a Final AD with Request for Comments, postponing the public consultation process until after publication. 	
	 Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <u>ADs@easa.europa.eu</u>. 	
	 For any question concerning the technical content of the requirements in this AD, please contact: <u>Solo Kleinmotoren GmbH</u>, Postfach 60 01 52, D 71050 Sindelfingen, Germany, Telephone: +49 07031-301-0, Fax: +49 07031-301-136, E-mail <u>aircraft@solo-germany.com</u>. 	